## Aircraft measurements of anthropogenic NO<sub>2</sub> with an imaging DOAS instrument

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## **Objectives**

- **Objectives**: Measuring the amounts and spatial distribution of tropospheric trace gases, e.g. for NO<sub>2</sub> pollution mapping, identification of source regions and source strengths, satellite data validation. **Why aircraft measurements and imaging DOAS?**
- Aircraft observations form a link between ground based observations and satellite measurements.
- Good spatial resolution ~ 100 m, at useful spatial coverage.
- Several viewing directions are observed at the same time, i.e. a broad stripe below the aircraft.

Advantage: less data is lost as cp. to scanning instruments, adjacent regions are viewed simultaneously.
 Status of the IUP Bremen iDOAS: The imaging DOAS instrument has been developed in 2011. Laboratory measurements have been performed to characterise the instrument, first test flights have been conducted during a flight campaign in June 2011. Latest work covers the determination of the corrected ground coordinates.

## The Polar-5 aircraft

AWI, Germany

Kenn Borek Air Ltd., Canada

Imaging

≤́ 200

g 300

#### Polar-5

Owner:

**Operator:** 

Registration Aircraft Type Length/Height/Span Speed Altitude (no pressurised cabin)

C-GAWI Basler BT-67 / DC3 21 m / 5.2 m / 29 m 50-105 m/s 100-19000 ft



Polar-5 in Hangar at Bremerhaven Luneort airport

## **Instrumental setup**

#### **Technical information and special features**

Spectral window/resolution:	420 - 460nm / 0.7-1.0nm
CCD Detector size:	512x512 pixels, 8.2x8.2 mm <sup>2</sup>
Detector type:	Frame transfer (FT) technique
Entrance optics:	Wide angle objective + fibre bundle
Fibre bundle:	35 sorted fibres imaged onto CCD

This special setup allows gap-free measurements (due FT CCD) and flexible positioning in aircraft (due to sorted fibre bundle).



#### **Observation and viewing geometry**

• Two nadir ports: spectrometer objective and picture camera

- Geolocation information: from GPS sensor and gyrometer
- Viewing directions: max. 35 LOS (line of sight) from 35 fibres
- LOS after averaging across track: fibres combined to 9 LOS ( $\theta_i$ )
- Field of view: ~48° across track ( $\gamma$ ), ~3° along track ( $\theta$ )
- Swath width: on the order of flight altitude H
- Exposure time t<sub>exp</sub>: typ. 0.5s
- Measurement sequence: no gaps between exposures
  Spatial resolution: ~100 m (at H=1km flight altitude, 9 viewing directions, depending on flight altitude and required SNR)





DOAS instrument



Nadir viewing ports: spectrometer objective (left) & picture camera (right)

#### **Measurement example**

rack in the Polar-5 aircraft

Spectrometer CCD image

#### Picture camera



The blue box (right) marks the field of view of the spectrometer. Bright scenes such as roads are identified by higher intensity in single LOS on the CCD, demonstrating good spatial imaging.

## Observation of NO<sub>2</sub> columns with corrected geolocation

#### NO<sub>2</sub> retrieval above a power plant

Black coal power plant (848 MW) at Ibbenbüren (52° 17.2' N, 7° 44.8' E)
The slant column of NO<sub>2</sub> is retrieved by DOAS (see settings below)
Large variation of NO<sub>2</sub> amounts across and along track are observed
The NO<sub>2</sub> in the exhaust plume of the power plant is clearly visible



 $NO_2$  amounts along the flight track retrieved from the flight on 04.06.2011. Downwind from the power plant of Ibbenbüren, strong enhancement of  $NO_2$  is visible (average wind direction was about East North-East). Enhanced  $NO_2$  is on the order of  $10^{16}$  molec/cm<sup>2</sup>.

#### Computation of viewing geometry in flight

- Calculation of correct ground geolocation from the field of view is important
- Consideration of the aircraft angles (pitch, roll and yaw) is required in addition to GPS position
- The ground pixel centre and corner coordinates for each LOS are calculated for the start and end of each exposure to determine the pixel area



#### Influence of aircraft angles on geolocation

Aircraft movement changes the ground pixel geolocation significantly. The magnitude of the displacement additionally depends on flight altitude and LOS.

Displacements of the ground pixel at a flight altitude ~1km:

- during straight tracks: on the order of a few tens of metres
- during turns: typically on the order of 500 m with maximum distances larger than 1km
- average for central flight pattern on 04.06.11: around 160m



Flight track in black from GPS latitude and longitude data. The coloured track shows the roll angle within the corrected ground pixel area from the three central LOS. Positive turns (right wing down) are shown in red, negative turns are plotted in blue. Displacements during curved parts of the track are clearly seen.

## NO<sub>2</sub> retrieval settings

#### **Retrieval Settings**

## **Summary and Outlook**

Summary

### Fitting window Trace gases Atmospheric effects Polynomial Reference I<sub>0</sub>

425 – 450 nm

NO<sub>2</sub> (293K), O<sub>3</sub> (241K), O<sub>4</sub> (296K), H<sub>2</sub>O (HITRAN) Ring (SCIATRAN calculated), intensity offset quadratic

from same LOS, rural scene, low NO<sub>2</sub> content, 1 min. average individual slit function for each viewing direction

#### **Detection Limit**

Slit function

Depends on integration time (typ. 0.5 s exposure time, binning leads to slightly larger pixels)
 For 2s, the detection limit lies around 10<sup>15</sup> molec/cm<sup>2</sup>, optical density rms is on the order of 10<sup>-3</sup>
 → Trade-off between ground spatial resolution and signal-to-noise ratio

# Universität Bremen

The newly developed airborne imaging DOAS instrument performed well during test flights in June 2011
Good imaging quality; constant slit function along spectral axis; promising measurement quality
NO<sub>2</sub> column amounts have been retrieved, pollution point sources can be observed
Aircraft pitch, roll and yaw angles are fully taken into account, for correct ground geolocation
Noise on the NO<sub>2</sub> amounts is quite large, data averaging may further improve SNR

#### Activities for the future

• Accurate consideration of air mass factor needs to be implemented

• Emission strength of sources may be determined and further dedicated campaigns will be conducted

#### **Selected References**

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#### Acknowledgements

The authors gratefully acknowledge financial support by the University of Bremen and by ESA through the TIBAGS project. Campaign support from AWI Bremerhaven, Martin Gehrmann and Franziska Nehring, is gratefully acknowledged. Thank you to the aircraft crew from Kenn Boreck, Canada.

## www.doas-bremen.de